

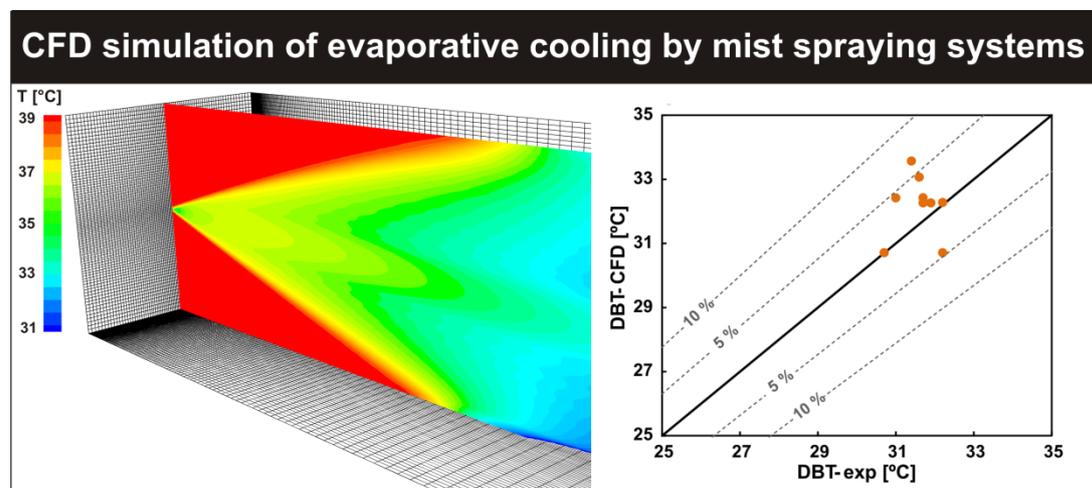
## Evaporative cooling by water spray systems: CFD simulation, experimental validation and sensitivity analysis

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### Graphical abstract



### Research highlights:

- CFD simulation of evaporative cooling by mist spraying systems.
- Grid-sensitivity analysis and validation with wind-tunnel measurements.
- Lagrangian-Eulerian approach can accurately simulate the evaporation process.
- Maximum local deviations: 10% for dry bulb temperature, 5% for wet bulb temperature and 7% for specific enthalpy.

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## Evaporative cooling by water spray systems: CFD simulation, experimental validation and sensitivity analysis

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### Abstract

Evaporative cooling by water spray is increasingly used as an efficient and environmentally-friendly approach to enhance thermal comfort in built environments. The complex two-phase flow in a water spray system is influenced by many factors such as continuous phase velocity, temperature and relative humidity patterns, droplet characteristics and continuous phase-droplet and droplet-droplet interactions. Computational Fluid Dynamics (CFD) can be a valuable tool for assessing the potential and performance of evaporative cooling by water spray systems in outdoor and indoor urban environments. This paper presents a systematic evaluation of the Lagrangian-Eulerian approach for evaporative cooling provided by the use of a water spray system with a hollow-cone nozzle configuration. The evaluation is based on grid-sensitivity analysis and validated using wind-tunnel measurements. This paper also presents a sensitivity analysis focused on the impact of the turbulence model for the continuous phase, the drag coefficient model, the number of particle streams for the discrete phase and the nozzle spray angle. The results show that CFD simulation of evaporation by the Lagrangian-Eulerian (3D steady RANS) approach, in spite of its limitations, can accurately predict the evaporation process, with local deviations from the wind-tunnel measurements within 10% for dry bulb temperature, 5% for wet bulb temperature and 7% for the specific enthalpy. The average deviations for all three variables are less than 3% in absolute values. The results of this paper are intended to support future CFD studies of evaporative cooling by water spray systems in outdoor and indoor urban environments.

**Keywords:** Computational Fluid Dynamics (CFD), Sensitivity study, Validation, Urban physics, Evaporation, Hollow-cone spray

### 1. Introduction

As a result of climate change more buildings will be exposed to milder winters and hotter summers [1,2]. Research indicates that a major European heat wave, such as that of 2003, will occur more frequently in the future [3] and it could become a common event by 2040 [4]. Increased heat waves and heat stress are likely to cause increased illness and death as occurred in the hot summers of 2003 and 2006 [5]. These problems are aggravated by the urban heat island effect (UHI) [6,7]. The term urban heat island is used for urban areas which exhibit higher temperatures than their rural surroundings. Therefore, adaptation strategies such as evaporative cooling need to be evaluated and implemented to reduce heat stress in the outdoor and indoor urban environment.

Several research organizations and consortia have initiated projects regarding climate change adaptation in cities as the Intergovernmental Panel on Climate Change (IPCC) has expressed the importance of adaptation measures [8]. Climate Proof Cities (CPC) is one of these research consortia investigating the climate vulnerability of urban areas and the development of climate change adaptation measures [9]. The consortium consists of universities, research institutes, policy makers and city officials to perform both basic and applied science, the latter being an integrated and thorough analysis for several locations in the Netherlands.

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Evaporative cooling by water spray is increasingly used as an efficient and environmentally-friendly approach to enhance thermal comfort in urban environments (outdoors and indoors) (e.g. [10,11]). In a water mist spray, a cloud of very fine water droplets is produced using atomization nozzles. It enhances mixing and increases the contact surface area between the air stream and the water droplets resulting in a higher rate of evaporation yielding greater cooling of the ambient air.

For assessing the potential and performance of evaporative cooling by water spray systems in outdoor and indoor environments, different methods can be used: (i) full-scale measurements, (ii) wind-tunnel measurements, and (iii) numerical simulation with Computational Fluid Dynamics (CFD). Full-scale measurements offer the advantage that the real situation is studied and the full complexity of the problem is taken into account. However, full-scale measurements are usually only performed in a limited number of points in space. In addition, there is no or limited control over the boundary conditions. Reduced-scale wind-tunnel measurements allow a strong degree of control over the boundary conditions, however at the expense of – sometimes incompatible – similarity requirements. Furthermore, wind-tunnel measurements are usually also only performed in a limited set of points in space. CFD on the other hand provides whole-flow field data, i.e. data on the relevant parameters in all points of the computational domain [12–15]. Unlike wind-tunnel testing, CFD does not suffer from potentially incompatible similarity requirements because simulations can be conducted at full scale. CFD simulations easily allow parametric studies to evaluate alternative design configurations, especially when the different configurations are all a priori embedded within the same computational domain and grid (see e.g. [14]). However, the accuracy and reliability of CFD are of concern, and verification and validation studies are imperative (e.g. [13,16–19]). CFD is increasingly used to study a wide range of atmospheric and environmental processes (e.g. [13,20–22]). Examples include pedestrian wind comfort and wind safety around buildings [23–26], natural ventilation of buildings [12,14,27–35], air pollutant dispersion [36–39], wind-driven rain [40] and convective heat transfer [41,42]. CFD has also been used on several occasions in the past to evaluate the performance of spray systems for different applications (e.g. [43–48]). In the vast majority of these studies the Lagrangian-Eulerian (LE) approach has been used in which the continuous phase (air in this study) is represented in an Eulerian reference frame while the discrete phase (water droplets in this study) is represented in a Lagrangian reference frame. The numerical implementation of this approach was introduced and applied by O'Rourke [49,50] and Dukowicz [51] for internal combustion engine applications. However, it has been developed and used for many other applications including evaporating spray systems. A comprehensive review of the LE method including its advantages over the Eulerian-Eulerian (EE) method, modelling issues and numerical implementation is provided by Subramaniam [52].

To the best of our knowledge, a detailed evaluation of the LE approach for predicting evaporative cooling has not yet been performed. This paper presents a systematic evaluation of the LE approach for predicting evaporative cooling provided by a water spray system with hollow-cone nozzle. The evaluation is based on grid-sensitivity analysis and on validation with wind-tunnel measurements by Sureshkumar et al. [53]. This paper also presents a sensitivity analysis focused on the impact of the turbulence model for the continuous phase and the number of particle streams for the discrete phase. In addition, the important impact of nozzle spray angle is demonstrated.

The results of this paper are intended to support future CFD studies of evaporative cooling by water spray systems in outside and inside urban environments.

## 2. Wind-tunnel experiments

In the experiments by Sureshkumar et al. [53] the evaporative cooling performance of a hollow-cone nozzle spray system was investigated. The experiments were performed in an open-circuit wind-tunnel with a uniform mean wind speed. The test section of the wind tunnel was 1.9 m long with a cross section of 0.585 m × 0.585 m (Fig. 1a). The dry-bulb temperature (DBT) and wet-bulb temperature (WBT) variations of the air stream between the inlet plane of the test section, where the spray nozzle was installed, and its outlet plane were measured for different air flow conditions and spray characteristics.

The inlet air DBT and WBT were measured by using two thermocouples placed upstream of the nozzle. Electric heaters were employed upstream of the tunnel blower to reduce the impacts of the background air temperature fluctuations. These fluctuations were limited within  $\pm 0.3$  °C during each set of experiments. The outlet air DBT and WBT were measured using 18 thermocouples installed at the tunnel outlet (Fig. 1b). A thermal probe installed upstream of the spray nozzle was used to measure the air stream velocity. The maximum experimental uncertainty for the mean velocity was estimated to be less than  $\pm 0.05$  m/s for air velocity up to 2 m/s and  $\pm 0.2$  m/s for air velocity between 2 and 4 m/s.

A drift eliminator with z-shaped plates was placed close to the tunnel outlet to collect the remaining water droplets in the air flow to avoid wetting of the thermocouples. The sump water was collected in a separate tank to avoid mixing of supply and sump water in order to keep the water inlet temperature constant during each set of experiments. The inlet and outlet water temperatures were measured using two thermocouples upstream of the nozzle and downstream of the drift eliminator, respectively. Water pressure was also measured by a pressure gauge upstream of the nozzle.

In order to evaluate the impact of nozzle characteristics on cooling performance of the spray system, four identical nozzles but with different discharge openings of 3, 4, 5 and 5.5 mm were used. Each nozzle was installed in the middle of the test section (Fig. 1a) and designed in a way that the exiting water forms a hollow-cone sheet disintegrating into droplets. The droplet diameter distribution was determined using an image-analysing technique. The uncertainty of this technique for the mean droplet size was estimated to be  $\pm 22\%$ . The half-cone angle was measured in still air and reported as a function of nozzle diameter, water pressure and background wind speed. As the exact value of the half-cone angle was not reported by Sureshkumar et al. [53], in Sec. 5.4 the influence of this parameter will be investigated. No correlations between droplet size and velocity were given by Sureshkumar et al [53].

The experiments were conducted in two periods of time to resemble different ambient (i.e. in the wind tunnel) conditions. The first set of experiments, which is used in the present study, were carried out in April-June representing a hot and dry climate condition in which DBT and humidity ratio (RH) ranged between 35 and 45 °C, and 10 and 35%, respectively. The second set was carried out in July-September representing hot-humid conditions with ambient DBT between 25 and 40 °C and RH between 30% and 90%. The inlet water temperature varied between 33 and 36 °C for the two ambient conditions. For each ambient condition, experiments were conducted for 36 cases; four different nozzle discharge diameters (i.e. 3, 4, 5 and 5.5 mm), three inlet nozzle gauge pressures (1, 2 and 3 bar) and three background wind speeds (1, 2 and 3 m/s). The three cases with a nozzle discharge diameter of 4 mm and a gauge pressure of 3 bar were taken since droplet size distribution data were also available for these cases.

### 3. CFD simulation

In this study the commercial software ANSYS/Fluent 12.1 [54] is used in which the Lagrangian-Eulerian approach is implemented to simulate multi-phase flows in sprays and atomizers.

#### 3.1 Computational geometry and grid

A computational model was made of the wind-tunnel test section with dimensions 0.585 m  $\times$  0.585 m  $\times$  1.9 m (Fig. 2a). Geometry and grid generation was executed with the pre-processor Gambit 2.4.6, resulting in a grid with 1,018,725 hexahedral cells (Fig. 2b). A stretching ratio of 1.05 controls the cells located in the immediate surroundings of the nozzle. The grid resolution resulted from a grid-sensitivity analysis that will be outlined in section 5.1. The minimum and maximum cell volumes in the domain are approximately  $1.9 \times 10^{-8} \text{ m}^3$  and  $5.9 \times 10^{-6} \text{ m}^3$ , respectively. The distance from the centre point of the wall adjacent cell to the walls is 0.006 m. This corresponds to  $y^*$  values between 35 and 135 for the case with the maximum air inlet velocity (i.e. 3 m/s). As standard wall functions are used in this study, these values ensure that the centre point of the wall-adjacent cell is placed in the logarithmic region of the boundary layer.

#### 3.2 Boundary conditions

In the simulations, the mean velocity inlet boundary condition for the continuous phase is a uniform profile according to the measured data ( $=U_\infty$ ). As the turbulence characteristics of the flow were not reported by Sureshkumar et al [53], a turbulence intensity,  $I$ , of 10% is assumed for the inlet flow. The turbulent kinetic energy  $k$  is calculated from  $U_\infty$  and  $I$  using Eq. (1). The turbulence dissipation rate,  $\varepsilon$ , is given by Eq. (2) where  $C_\mu$  is a constant ( $\sim 0.09$ ). The turbulence length scale,  $l$ , in this equation is taken as  $l = 0.07D_H$  where  $D_H$  is the hydraulic diameter of the domain which is equal to the width of the test section ( $= 0.585 \text{ m}$ ).

$$k = (U_\infty \cdot I)^2 \quad (1)$$

$$\varepsilon = C_\mu^{3/4} \frac{k^{3/2}}{l} \quad (2)$$

The thermal boundary condition at the inlet is a constant temperature. A fixed vapour mass fraction is also calculated based on the experimental data and imposed at the inlet as a boundary condition for the vapour transport equation. Note that the vapour mass fraction for the moist air can be taken as  $x/(x+1)$  where  $x$  ( $\text{kg}_{\text{vapour}}/\text{kg}_{\text{dry-air}}$ ) is the humidity ratio of air. The walls of the computational domain are modelled as no-slip walls with zero roughness height  $k_s = 0$ . The standard wall functions [55] are applied. The adiabatic thermal boundary condition is used for these surfaces. Zero static gauge pressure is applied at the outlet plane.

Special attention is needed for the discrete phase boundary conditions to take the effect of the wind-tunnel walls into account. The phenomena occurring when droplets impinge on solid surfaces are complicated and depend on the physical properties of the droplets such as surface tension, viscosity, density, temperature and diameter, and on impingement conditions such as impact angle and velocity of droplets relative to the wall [56,57]. In addition, many studies were carried out to investigate the influence of wall characteristics such as roughness, temperature and wettability of the surface (e.g. [58]). A detailed review of the studies on droplet-wall impact can be found in Moreira et al. [59]. Depending on the mentioned conditions, droplets may float and be lost in a liquid film, may be reflected or may disintegrate into smaller droplets. As the temperature of the wall in the experiments by Sureshkumar et al. [53] is less than the boiling temperature and the Leidenfrost temperature of the droplets, for the simulations it is assumed that the droplets are entrained in a water film along the walls after impingement [60]. In this case, the ‘‘reflected’’ boundary condition is used and it is assumed that after impingement the normal momentum component is zero while the tangential component remains the same. The presence of a liquid on the surface changes the boundary condition and the impact leads to a liquid-liquid interaction. In this case, the impact characteristics depend on the surface roughness but also the film thickness compared with the droplet size [61]. However, the effect of the film thickness is not taken into account.

As mentioned in section 2, a drift eliminator was used in the experiments near the outlet plane. In this study the drift eliminator is not included in the computational domain because a detailed description of its characteristics was not reported by Sureshkumar et al. [53]. However, the impact of such plates is taken into account by using the ‘‘escape’’ boundary condition at the outlet, assuming that the upstream impact of the drift eliminator on the air flow pattern is negligible. By using this boundary condition, droplets leave the domain with their current conditions (i.e. velocity, temperature and vapour mass fraction at the outlet plane) and trajectory calculations are terminated [54].

### 3.3 Droplet characteristics

Geometric properties of droplets in a spray system are of the most important factors that affect the dynamic behaviour of the droplets, but also the heat and mass transfer rates between the continuous and discrete phase. These geometric properties include size and shape of the droplets. In the experiments by Sureshkumar et al. [53], an image-analysing technique was used. Fig. 3a shows the discrete number density distribution for the case that the nozzle diameter and water pressure were 3 mm and 4 bar, respectively. The arithmetic mean diameter,  $D_{10}$ , and the Sauter mean diameter,  $D_{32}$ , are about 198 and 293  $\mu\text{m}$ , respectively, based on the representative drop diameters standardised by Mugele and Evans [62]:

$$D_{pq} = \left[ \frac{\int_0^\infty D^p f_N(D) dD}{\int_0^\infty D^q f_N(D) dD} \right]^{\frac{1}{p-q}} \quad (3)$$

where  $f_N(D)$  is the number density distribution and  $p$  and  $q$  are positive integers.

The droplet size distribution can be modelled in different ways, including semi-empirical (i.e. Rosin-Rammler [63] and Nukiyama-Tanasawa [64]), maximum entropy formalism (MEF) [65,66], discrete probability function (DPF) [67,68] and stochastic [69] models. Detailed discussions on methods for modelling drop size distribution along with their strengths and weaknesses can be found in the literature (e.g. [70,71]). The semi-empirical methods are the most common methods in which droplet size distribution is determined by fitting the measured data to pre-determined mathematical functions [70]. In the present work the Rosin-Rammler [63] model is used. This model assumes an exponential relationship between the droplet diameter,  $D$ , and the mass fraction of droplets with diameters greater than  $D$ , which can be expressed according to Eq. (4):

$$Y_D = e^{-(D/\bar{D})^n} \quad (4)$$

where  $Y_D$  is the mass fraction of droplets with diameters greater than  $D$ ,  $\bar{D}$  the mean diameter and  $n$  the spread parameter as an indicator of the distribution width. In this study, the following steps are taken to obtain the Rosin-Rammler distribution:

- 1)  $Y_D$  is obtained from the mass density distribution  $f_M(D)$ . The mass density distribution is calculated using Eq. (5) in which the number density distribution  $f_N(D)$  is related to  $f_M(D)$  [72]:

$$f_M(D) = \frac{N_0 m}{M_0} f_N(D) \quad (5)$$

where  $N_0$  and  $M_0$  are the total number and total mass of the sample droplets. In this equation,  $m$  is the mass of a single droplet of diameter  $D$  and density  $\rho$  (i.e.  $\pi\rho D^3/6$ ). The results are shown in Fig. 3b. The calculated values of  $Y_D$  for each droplet size group are provided in Fig. 4.

- 2) The mean diameter,  $\bar{D}$ , is determined from Fig. 4 for  $Y_D = e^{-1}$  (i.e. when the droplet diameter,  $D$ , equals  $\bar{D}$ ). For the current experimental data  $\bar{D}$  is 369  $\mu\text{m}$ .
- 3) The spread parameter,  $n$ , for each diameter group is derived from Eq. (4) (i.e.  $n = \ln(-\ln Y_D)/\ln(D/\bar{D})$ ). Averaging over these values yields the spread parameter for the Rosin-Rammler distribution, which is 3.67 for the current experimental data. This is within the range proposed by Lefebvre [73] for sprays, i.e.  $1.5 < n < 4$ . The Rosin-Rammler curve fit is shown in Fig. 4 (solid line).

In this study, the smallest droplet diameter to be considered in the size distribution of the Rosin-Rammler model is 74  $\mu\text{m}$ , corresponding to the minimum resolution of the droplet measurements. The largest droplet diameter is considered 518  $\mu\text{m}$ , based on the largest droplet diameter in the samples (Fig 3a). 20 diameters are also assumed to be injected from each droplet stream into the domain. This number was taken based on a sensitivity analysis (not shown in this paper).

The spherical drag law is used to estimate the drag coefficients acting on droplets. It assumes that the surface tension on the drop-fluid interface is strong enough to resist the tendency of the aerodynamic force to deform the drop ( $We \ll 1$ ) (e.g. [74]). In this scenario, droplets are assumed to be non-deforming spheres and drag coefficients ( $C_d$ ), as functions of the Reynolds number ( $Re$ ), are estimated based on experimental drag data for solid spheres. Various correlations for the drag coefficient of spherical droplets (particles) can be found in the literature (e.g. [75–78]). In the present study the correlation by Morsi and Alexander [77] is used. This correlation proposes the following drag coefficients for a wide range of Reynolds numbers up to  $5 \times 10^4$ :

$$C_d = \frac{K_1}{Re} + \frac{K_2}{Re^2} + K_3 \quad (6)$$

In which  $K_1$ ,  $K_2$  and  $K_3$  are three constants (Table 1).

In this study, the spray is assumed to be dilute and the impact of collision of droplets is not taken into account.

### 3.4 Spray nozzle characteristics

In this study the hollow cone spray model provided by Ansys/Fluent 12.1 [54] is used. The water droplets are injected into the computational domain from a nozzle with 4 mm diameter positioned in the middle of the inlet plane of the computational domain and oriented horizontally in downstream direction. The total mass flow rate and temperature of the injected water droplets are imposed according to the experimental data (Table 2). The total number of droplet (particle) streams is assumed to be 300 meaning that they are released from 300 uniformly-distributed points on the nozzle opening perimeter. In order to ensure that the results are independent of the number of droplet streams, the impact of this factor will be investigated in section 5.3. The sheet velocity,  $U_0$ , is used for the initial velocity of the droplet streams.  $U_0$  is calculated as  $C_v(2\Delta P/\rho_w)^{0.5}$  where  $C_v$  is the nozzle coefficient,  $\Delta P$  the pressure difference along the nozzle and supply pipe and  $\rho_w$  the water density. As recommended by Sureshkumar et al. [43] for the given nozzle  $C_v$  is approximately 0.9.

### 3.5 Solver settings

For the continuous phase flows, the 3D steady RANS equations for conservation of mass, momentum and energy are solved in combination with the realizable k- $\epsilon$  turbulence model by Shih et al. [79]. The SIMPLE algorithm is used for pressure-velocity coupling, pressure interpolation is second order and

second-order discretisation schemes are used for both the convection terms and the viscous terms of the equations.

Lagrangian trajectory simulations are performed for the discrete phase. The discrete phase interacts with the continuous phase, and the discrete phase model source terms are updated after each continuous phase iteration. To solve the equations of motion for the droplets, the Automated Tracking Scheme Selection is adopted to be able to switch between higher order lower order tracking schemes. This mechanism can improve the accuracy and stability of the simulations [52]. In this study, trapezoidal and implicit schemes are used for higher and lower order schemes, respectively. The solution of the droplet momentum, heat and mass transfer equations are solved in a fully coupled manner.

### 3.6 List of cases

Some main parameters of the three cases that are investigated in this study are given in Table 2. Case 3 is used for the sensitivity analysis.

## 4. Results

### 4.1 Grid-sensitivity analysis

A grid-sensitivity analysis was performed based on two additional grids; a coarser grid and a finer grid. An overall linear factor  $\sqrt{2}$  was used for coarsening and refining. The coarse grid had 360,000 cells, while the fine grid had 2,880,000 cells. The three grids are shown in Fig. 5. The DBT profiles along three lines at the outlet plane for the three grids are compared in Fig. 6. The concept of the grid-convergence index (GCI) by Roache is also used [80] (not shown in this figure). The results show a limited dependence of the DBT results on the grid resolution in the middle of the outlet plane with a maximum local GCI value of 0.17%. Negligible grid sensitivity is found for the other parts. In this case, the average discretisation error (GCI) for the lines on the left, middle and right side of the outlet plane is 0.05, 0.10 and 0.05%, respectively. Therefore, the basic grid was retained for further analysis.

### 4.2 Comparison of CFD results and wind-tunnel experiments

The CFD results for the three cases in Table 2 are compared with the wind-tunnel experiments by Sureshkumar et al. [53]. The comparison is performed for the DBT, WBT and specific enthalpy values in the nine measurement points. Note that the specific enthalpy of moist air,  $h$ , can be expressed as:

$$h = h_{dry.air} + x \cdot h_v \quad (7)$$

where  $h_{dry.air}$  is the specific enthalpy of dry air ( $\text{kJ}/\text{kg}_{dry.air}$ ) given by  $C_p \cdot T$ , where  $C_p$  is the specific heat capacity of air ( $\text{kJ}/\text{kg K}$ ).  $x$  is the humidity ratio ( $\text{kg}_{vapour}/\text{kg}_{dry.air}$ ) and  $h_v$  the specific enthalpy of water vapour. The results in Fig. 7 shows a good agreement, within 10% for DBT, 5% for WBT and 7% for the specific enthalpy for all cases. The exact reasons for these deviations are not clear, but they are probably caused by a combination of limitations of the LE approach and experimental uncertainties. Apart from the LE approach limitations, the impact of collision of droplets, droplets impingement on solid surfaces and the drift eliminator on the airflow are not considered into account in this study. For the experimental uncertainties, as mentioned by Sureshkumar et al. [53], the dominant uncertainty of the experiments is for the WBT measurements. According to Eq. 7, the deviation for  $h$  is highly dependent on the WBT deviations rather than the one for DBT. For example, for a case with DBT=39 °C and WBT=19 °C, 1°C of change in DBT and WBT results in about 0.06 and 6% deviations in  $h$ , respectively. This figure also indicates that the DBT increases by increasing the inlet velocity from 1 m/s (case 1) to 3 m/s (case 3). The higher inlet air velocity leads to a reduction in the residence time of droplets within the domain, resulting in less time for evaporation and higher outlet DBT. Note that the initial droplet velocities for the three cases are identical. However, the inlet DBT of the air in case 1 is higher than the other cases by more than 2 °C. The inlet WBT is approximately the same for the three cases (Table 2).

Fig. 8 shows the air speed (magnitude of 3D velocity), temperature and vapour mass fraction contours in the vertical centre plane. The cross-sectional profiles of these parameters along vertical lines on the centre plane of the domain are shown in Fig. 9. Both figures show that the air speed increases considerably close to the spray as the momentum lost by the droplets is obtained by the air. The liquid water concentration in the air is relatively large close to the nozzle and therefore momentum exchange is more pronounced in this area. Ghosh and Hunt [81] explains that, away from the nozzle the air and droplets behave almost independently concerning momentum exchange. In addition, from Fig. 8 and Fig. 9 the symmetric velocity, temperature and mass fraction distributions can be clearly observed close to the nozzle, which is extended until about half the length of the domain. The high

inertia of the droplets results in symmetric trajectories of the droplets close to the nozzle. However, farther downstream the nozzle, by the decreasing momentum of the droplets, gravity becomes relatively more pronounced, resulting in a downward movement of the droplets. This leads to asymmetric distributions of speed, temperature and vapour mass fraction. The larger number of droplets is in the lower part of the domain, results in more cooling effects in this area.

Fig. 10a shows how the non-dimensional droplet mass flow rate changes along the domain. A gradual decrease of the droplet mass flow rate is clearly observed. At the outlet of the domain, more than 3% of the initial droplet mass flow has evaporated. The droplet mass flow rate is consistent with the droplet mean diameters, shown in Fig. 10b. It can be seen that  $D_{10}$ , the arithmetic mean diameter and  $D_{32}$ , the Sauter mean diameter decrease monotonically with increasing the distance from the inlet. As in this study the effect of droplet collision and secondary break up are not taken into account, this reduction in the mean diameters is mainly caused by the evaporation effects. The overall reduction of  $D_{10}$  and  $D_{32}$  within the domain is 4 and 1  $\mu\text{m}$ , respectively.

In ANSYS/Fluent, the spray droplets are treated as parcels and each parcel includes a group of droplets, which all have the same characteristics such as temperature, diameter and speed. Fig. 11 shows the droplet (parcels) temperature and diameter distributions for three cross sections along the domain:  $x/L = 0.25, 0.5$  and  $0.75$ . At the inlet, droplets with different diameters are injected with the same temperature, according to experimental data (Table 2). Farther downstream, the temperature of the droplets decreases considerably. This reduction is highly dependent on the droplet diameters. For example, halfway the domain ( $x/L = 0.5$ ) the temperature reduction ranges from 13  $^{\circ}\text{C}$  for the smallest droplets to about 4  $^{\circ}\text{C}$  for the largest ones. Fig. 11 also shows that for  $x/L = 0.25$  and  $0.5$ , the temperature and diameter of the droplets with the same initial diameter change in an axisymmetric fashion. This is because of the symmetric flow field (Fig. 8) and symmetric droplet trajectories in the first half of the domain. A larger spread in the droplet temperatures can be seen at  $x/L = 0.75$ , because of the asymmetric behaviour of the airflow and droplets close to the outlet. The evaporation along the domain causes the temperature of the droplets to reduce gradually and reach minimum values at the outlet of the domain.

The difference between the sensible heat at different cross sections and the one at the inlet is presented in Fig. 12, as an indication of the evaporative cooling capacity of the spray system. The cooling increases gradually along the domain and reaches to about 8 kW at the outlet plane. The amount of cooling is related to the sensible heat convection from/to the droplets and the latent heat of evaporation. To gain insight into the performance of the spray system, the temperature difference needed to provide the same amount of cooling (i.e. 8 kW) with a sensible cooling process can be calculated using  $Q = \dot{m}C_p\Delta T$ . In this case the temperature difference is about 6.5  $^{\circ}\text{C}$ .

## 4.2. Sensitivity analysis

To analyse the sensitivity of the results to various computational and design parameters, systematic changes are made to the reference case (case 3), outlined in the previous section. In every sub-section, one of the computational and design parameters is varied, while all others are kept identical to those in the reference case.

### 4.2.1 Impact of turbulence models

3D steady RANS simulations of the continuous phase were made with five turbulence models: (1) the standard k- $\epsilon$  model (Sk- $\epsilon$ ) [82]; (2) the realizable k- $\epsilon$  model (Rk- $\epsilon$ ) [79]; (3) the renormalization Group k- $\epsilon$  model (RNG k- $\epsilon$ ) [83,84]; (4) the standard k- $\omega$  model (Sk- $\omega$ ) [85] and (5) the Reynolds Stress Model (RSM) [86]. For the standard k- $\omega$  model, the inlet vertical profile for  $\omega$  is determined from that of k and  $\epsilon$  using Eq. (8), where  $C_{\mu}$  is equal to 0.09. For the RSM model, the Reynolds stress components are obtained from the turbulent kinetic energy k assuming isotropy of turbulence (Eqs. (9) and (10)):

$$\omega = \frac{\epsilon}{C_{\mu}k} \quad (8)$$

$$\overline{u_i^2} = \frac{2}{3}k \quad (i = 1, 2, 3) \quad (9)$$

$$\overline{u_i u_j} = 0.0 \quad (10)$$

The effects of the turbulence models on the results of DBT, WBT and  $h$  are shown in Fig. 13. No turbulence model can be considered superior than the others.

#### 4.2.2 Impact of number of particle (droplet) streams

In ANSYS Fluent, the number of droplet streams is the number of locations along the nozzle perimeter where the parcels are injected into the computational domain. Although the injected droplet flow rate is independent of the number of streams, the less droplet streams, the lower the computational time that is required. However, a sufficiently large number of streams needs to be employed to ensure the accuracy of the results. Fig. 14 shows the influence of the number of streams on the numerical results. 10 streams lead to a high discrepancy between the CFD results and experimental data. In this case, the average deviations for DBT, WBT and  $h$  are 18.8, 5.1 and 6.5%, respectively. By increasing the number of streams from 10 to 100, these deviations decrease considerably and reach to 3.1, 2.2 and 2.7% respectively. For this study, approximately 100 streams are considered sufficient for CFD results that are nearly independent of the number of droplet streams.

#### 4.2.3 Impact of spray nozzle angle

The impact of the spray angle is investigated by comparing the CFD results for different half-cone angles. The results are provided in Fig. 15 for the angles ranging from  $16^\circ$  to  $24^\circ$ , according to the experimental data. An increase in the half-cone angle provides a better agreement between the experimental data and the CFD results for all parameters. Note that the exact value of the spray angle was not reported by Sureshkumar et al. [53].

### 5. Discussion

This paper has presented a detailed and systematic analysis of the Lagrangian-Eulerian approach for predicting evaporative cooling provided by a hollow-cone water spray system. It has analysed the possibilities and limitations of the LE method for assessing the performance of a water spray system. A detailed sensitivity study was performed, for the important computational parameters: grid resolution, turbulence model for the continuous phase, number of particle streams for the discrete phase, and spray nozzle angle was demonstrated. It is important to mention the main limitations of this study.

- 1) The comparisons have been performed based on measured air DBT and WBT at the outlet of the domain.
- 2) The impacts of droplet deformation and secondary breakup have not been taken into account.
- 3) The spray was assumed to be dilute and the impact of collision of droplets is not taken into account.

Further research is necessary to validate the performance of the LE approach in predicting droplet characteristics such as velocity, temperature and diameter. Different droplet deformation models and collision models also need to be investigated.

The validation study and sensitivity analysis provide confidence in the adopted approach and guidelines for accurate simulation of evaporative cooling by water spray systems. This is intended to support future CFD studies of evaporative cooling by water spray systems in outdoor and indoor urban environments. Future research will focus on CFD simulations of evaporative cooling as an adaptation measure for climate change in a real complex case study. For this purpose, water spray systems will be implemented near buildings and in public spaces (streets). CFD modeling will be employed to perform a parametric analysis for different air flow conditions and spray characteristics to evaluate the performance of the spray systems in mitigating the heat stress in the outdoor environment. The thermal comfort will be assessed with the Universal Thermal Climate Index (UTCI) [87].

### 6. Conclusion

This paper presents a systematic evaluation of the Lagrangian-Eulerian approach for predicting evaporative cooling provided by a water spray system. This work was motivated by lack of knowledge on the accuracy and reliability of CFD for determining evaporative cooling provided by water spray systems. The evaluation is based on a grid-sensitivity analysis and on validation with wind-tunnel measurements by Sureshkumar et al. [53].

The present study showed that CFD simulation of evaporation by using the Lagrangian-Eulerian (3D steady RANS) approach, in spite of its limitations, can accurately predict the evaporation process with an acceptable accuracy. The local deviations from the wind-tunnel measurements are within 10% for dry bulb temperature, 5% for wet bulb temperature and 7% for the specific enthalpy. The average deviations for all three variables are less than 3% in absolute values. The impact of the turbulence

model for the continuous phase, the number of particle streams for the discrete phase and the half-cone angle have also been investigated, and it was demonstrated that the selection of these parameters is very important for accurate and reliable results.

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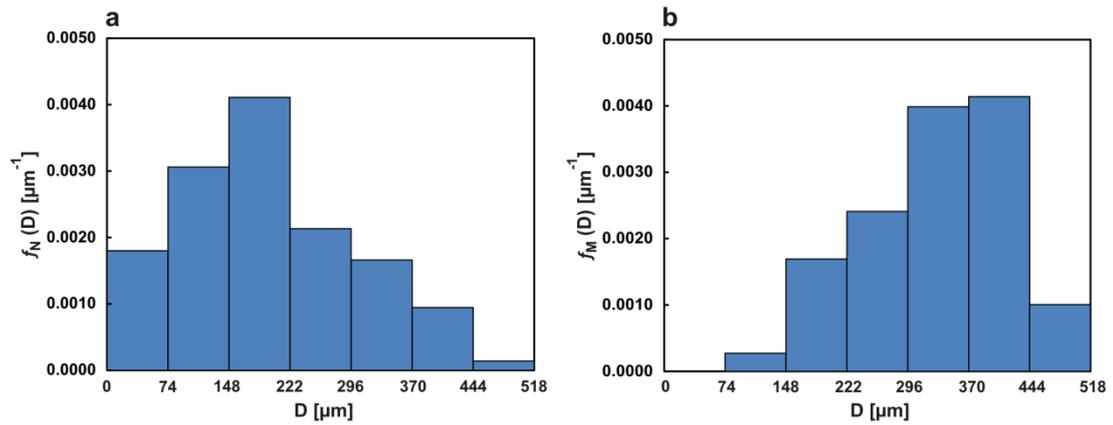
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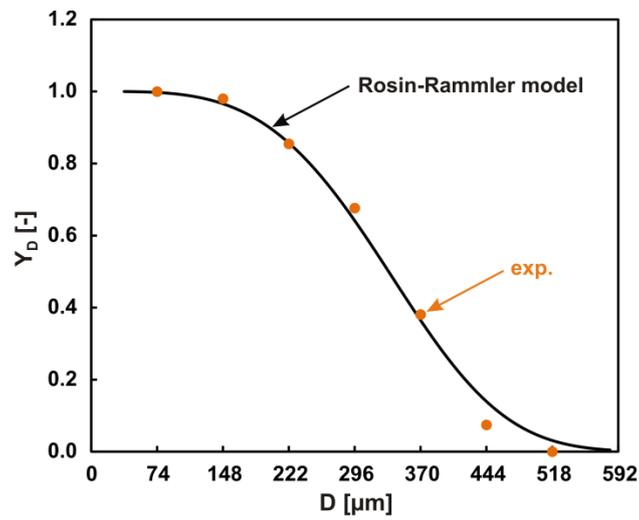
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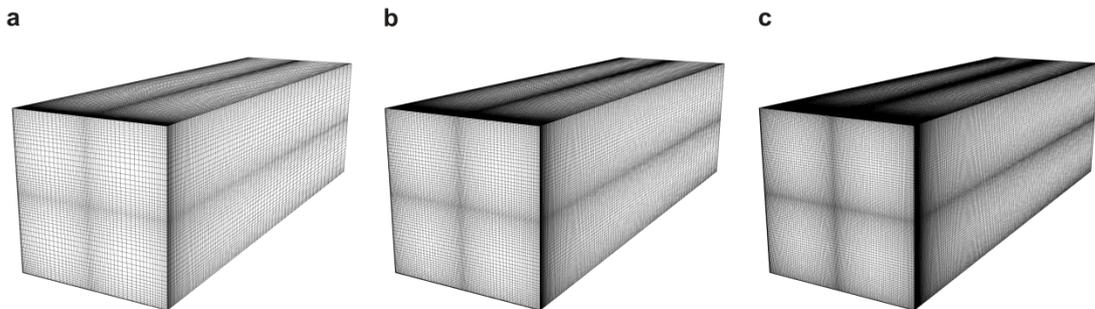




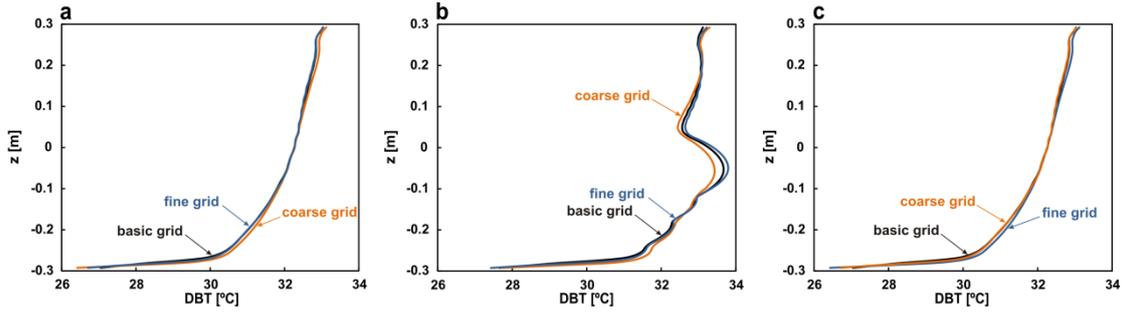
**Figure 3.** (a) Discrete number density distribution for the case with nozzle diameter of 3 mm and water pressure of 4 bar [53]. (b) Mass density distribution obtained from Eq. (5).



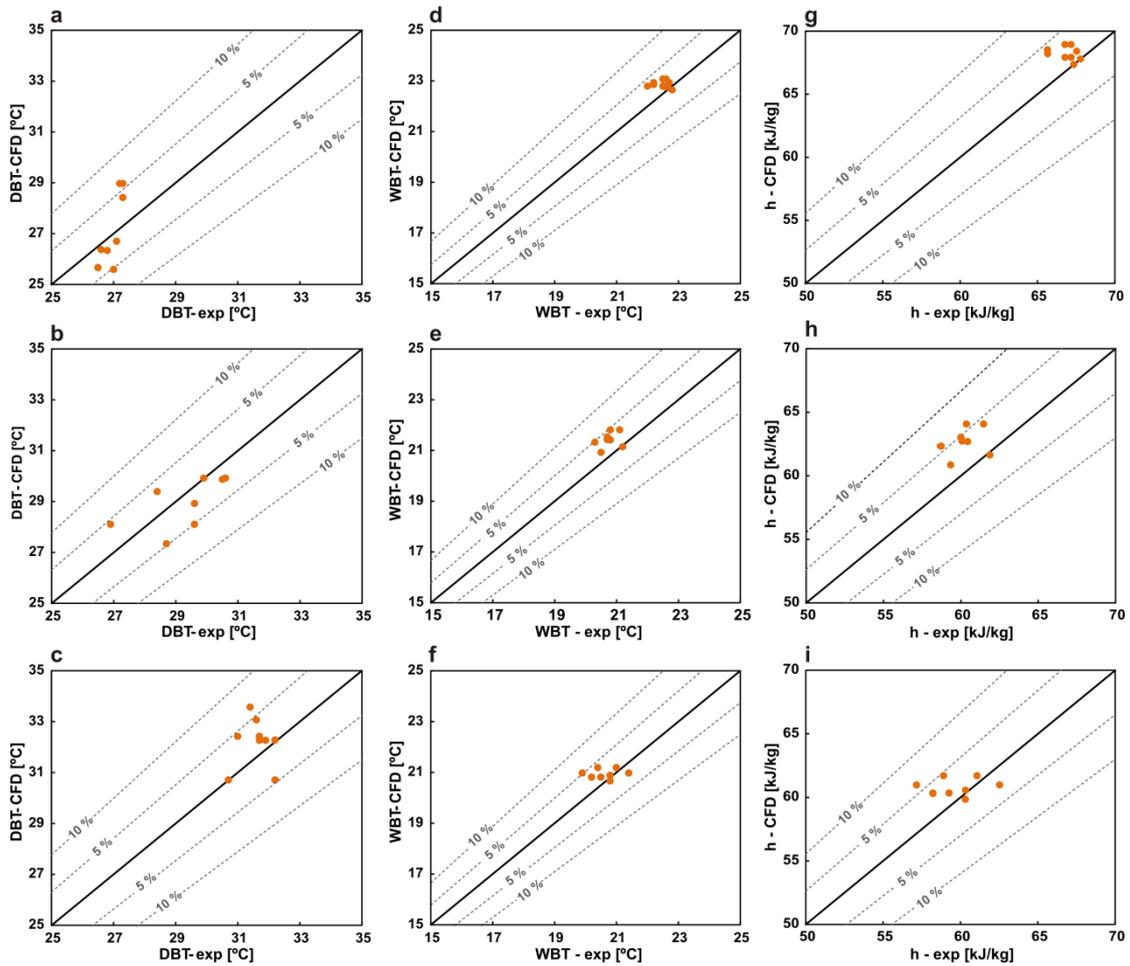
**Figure 4.** Rosin-Rammler curve fit (solid line) and experimental data of  $Y_D$  (dots).



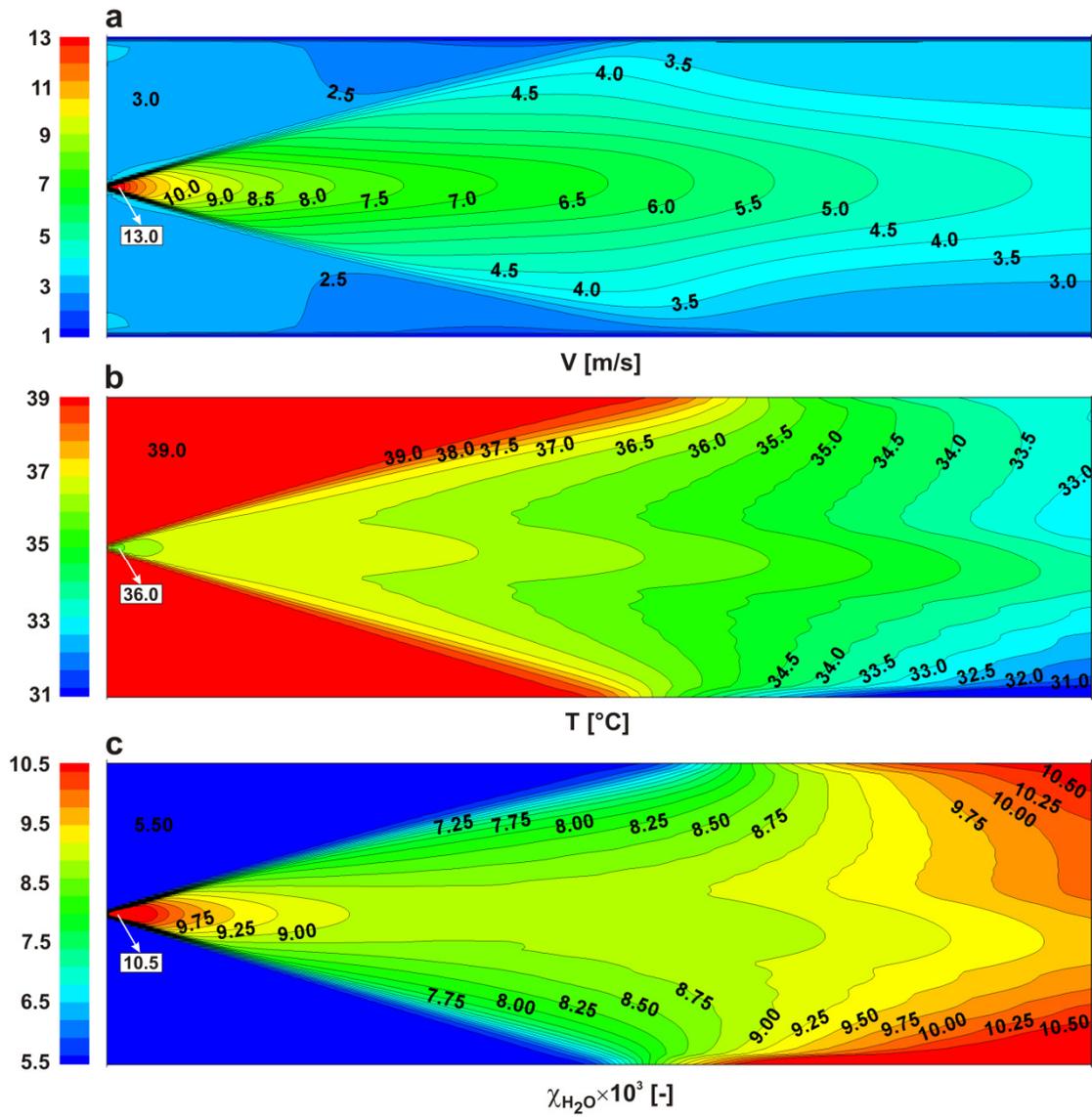
**Figure 5.** Computational grids for grid-sensitivity analysis. (a) Coarse grid (360,000 cells); (b) basic grid (1,018,725 cells) and (c) fine grid (2,880,000 cells).



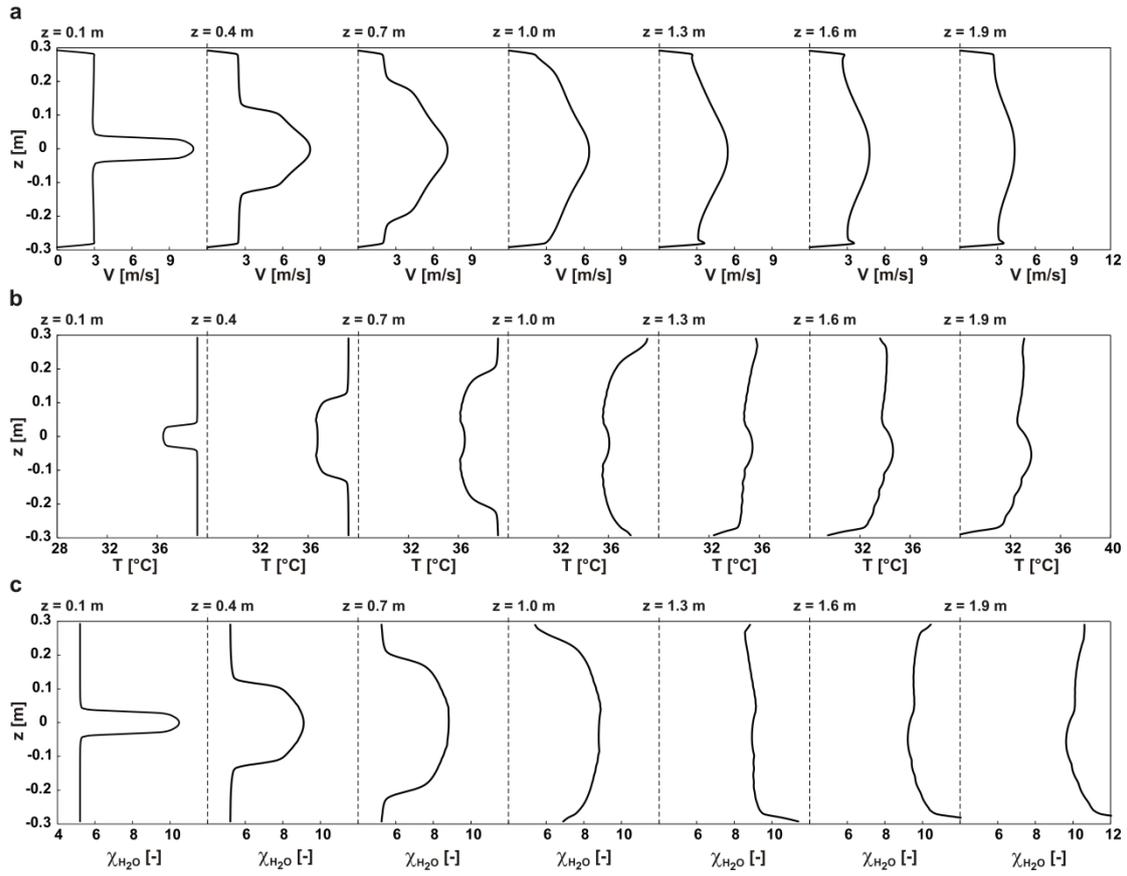
**Figure 6.** Results for grid-sensitivity analysis: DBT values along vertical lines (a) left, (b) middle and (c) right side of outlet plane for three grids.



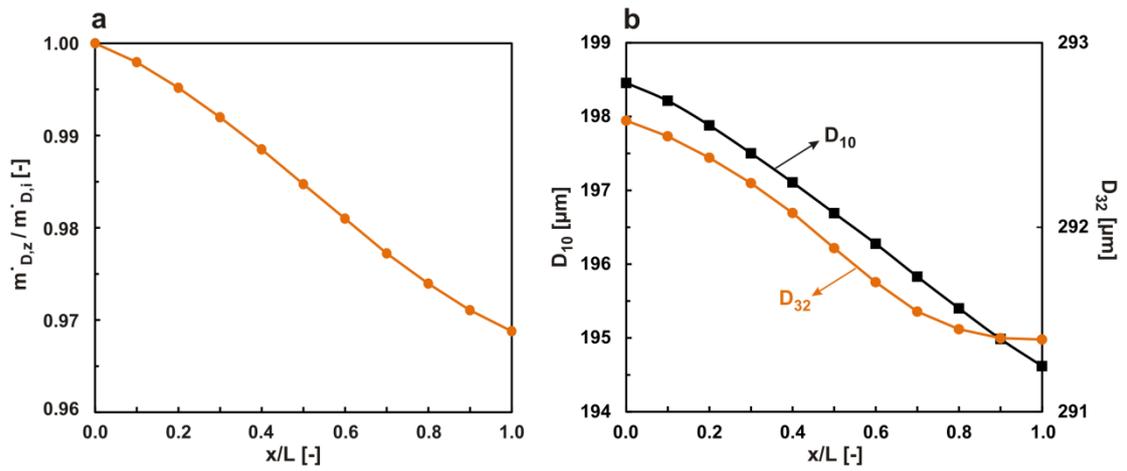
**Figure 7.** Comparison of calculated (CFD) and measured (exp. [53]) (a-c) DBT, (b-f) WBT and (g-i) specific enthalpy for case 1, 2 and 3, respectively.



**Figure 8.** (a) Air velocity; (b) temperature and (c) vapour mass fraction distribution in cross-section (centre plane) for case 3.



**Figure 9.** (a) Air speed; (b) temperature and (c) vapour mass fraction along vertical lines in the centre plane.



**Figure 10.** Profiles of (a) normalized droplet mass flow rate and (b) arithmetic ( $D_{10}$ ) and Sauter ( $D_{32}$ ) mean diameters along the domain for case 3.

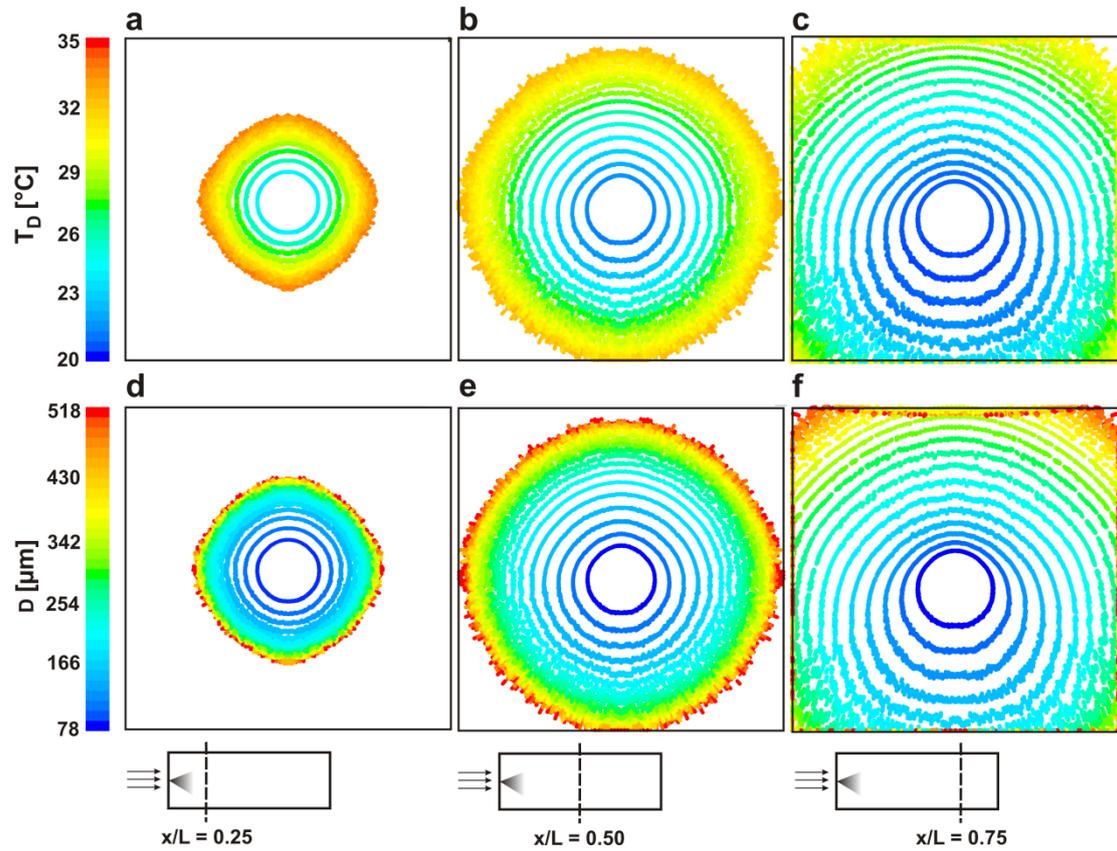


Figure 11. Droplet temperature distribution along the domain for three cross sections: (a)  $x/L = 0.25$ , (a)  $x/L = 0.50$ , (a)  $x/L = 0.75$ . (d-f) same for droplet diameter distribution.

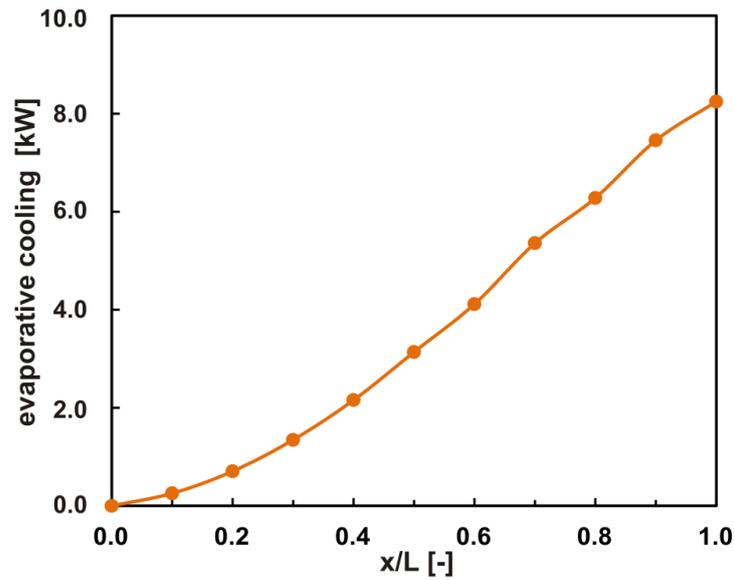
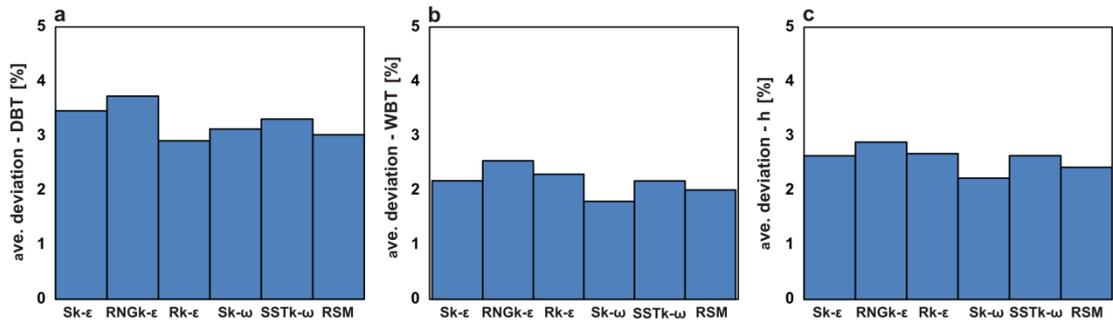
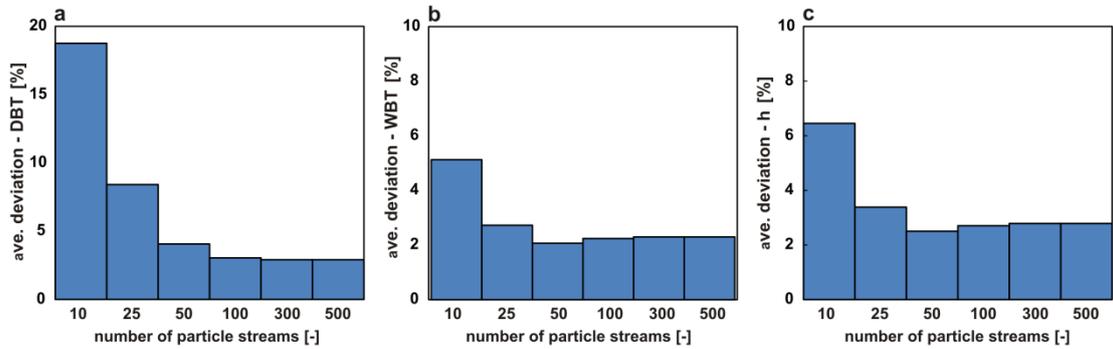


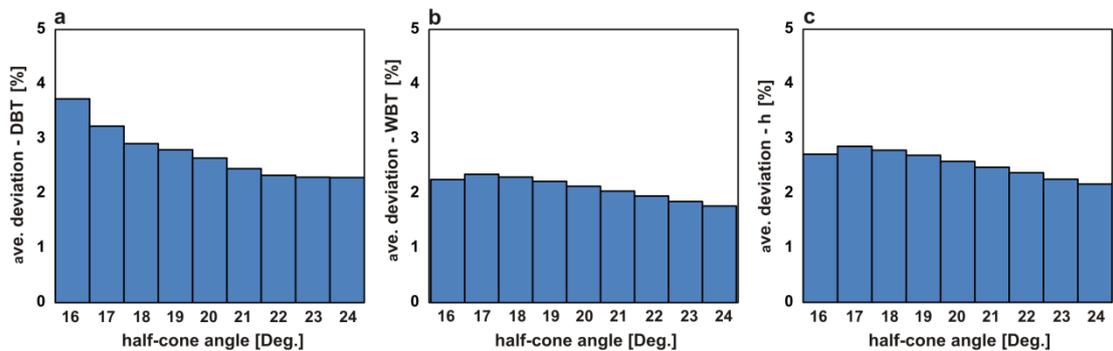
Figure 12. Profile of evaporative cooling capacity along the domain for case 3.



**Figure 13.** Impact of turbulence model on CFD simulation results of (a) DBT, (b) WBT and (c) specific enthalpy at the outlet plane (nine measurement points).



**Figure 14.** Impact of number of particle streams on CFD simulation results of (a) DBT, (b) WBT and (c) specific enthalpy at the outlet plane (nine measurement points).



**Figure 15.** Impact of cone angle on CFD simulation results of (a) DBT, (b) WBT and (c) specific enthalpy at the outlet plane (nine measurement points).

## TABLES

**Table 1.** Drag coefficient model constants for different Reynolds numbers [75].

Re range	$K_1$	$K_2$	$K_3$
$Re < 0.1$	24.00	0.00	0.00
$0.1 < Re < 1$	22.73	0.09	3.69
$1 < Re < 10$	29.17	-3.89	1.22
$10 < Re < 100$	46.50	-116.67	0.62
$100 < Re < 1000$	98.33	-2778.00	0.36
$1000 < Re < 5000$	148.62	-47500.00	0.36
$5000 < Re < 10000$	-490.546	578700.00	0.46
$10000 < Re < 50000$	-1662.50	5416700.00	0.52

**Table 2.** List of some main parameters of the cases.

case	Inlet air			Water				Spray nozzle	
	V (m/s)	DBT (°C)	WBT (°C)	P (bar)	T <sub>in</sub> (°C)	T <sub>out</sub> (°C)	$\dot{m}$ (lit/min)	D (mm)	$\alpha$ (deg.)
1	1	41.4	18.9	3	35.1	25.3	12.5	4.0	22.0
2	2	39.1	18.5	3	35.0	25.2	12.5	4.0	20.0
3	3	39.2	18.7	3	35.2	26.1	12.5	4.0	18.0